Fatigue Failure

Occurs when stresses are changing throughout the life of a part.

_____ starts with a crack that propagates until a catastrophic failure occurs.

This usually begins at a manufacturing defect or stress concentration that is subjected to tensile stresses for part of its load cycle.

Fatigue-Failure Models

- Stress-Life cyclic stresses are kept below fatigue strength or endurance limit. Most widely used.
- Strain-Life complicated not often used.

- sometimes used, gaining favor especially in the aircraft and aerospace industry.

Fatigue Stresses



Fully Reversed Failure Criteria

- Most data comes from R. R. Moore rotating-beam test.
- Highly polished specimen of 0.3 inches in diameter is subjected to pure bending stresses that are alternated by rotating the specimen.
- Rotation is at 1725 rpm.
- Takes 1/2 day to reach 10⁶ cycles.

S-N Curve for Steel with Sut = 100 ksi



Endurance Limit and Fatigue Strength

Endurance Limit of a Material (S_{e'}) stress below which fatigue failure does not occur regardless of the number of stress cycles.

<u>Strength</u> of a Material $(S_{f'})$ -stress below which fatigue failure does not occur for a specified number of stress cycles.

Typical S-N Curve for Non-Ferrous Metals



Estimating $S_{e'}$ or $S_{f'}$

For $S_{e'} = 0.5S_{ut}$ for $S_{ut} < 200$ ksi (1400 MPa) $S_{a'} = 100 \text{ ksi} (700 \text{ MPa})$ for S_{ut}>200 ksi (1400 MPa) For $S_{e'} = 0.4S_{ut}$ for $S_{ut} < 60$ ksi (400 MPa) $S_{e'} = 24 \text{ ksi} (160 \text{ MPa})$ for S_{ut} >60 ksi (400 MPa)

Estimating $S_{e'}$ or $S_{f'}$ cont'd

For _____: $S_{f'@5E8} = 0.4S_{ut}$ for $S_{ut} < 48$ ksi (330 MPa) $S_{f'@5E8} = 19$ ksi (130 MPa) for $S_{ut} > 330$ ksi (330 MPa)

For _____:

$$S_{e'} = 0.4S_{ut}$$
 for $S_{ut} < 40$ ksi (280 MPa)
 $S_{e'} = 14$ ksi (100 MPa)
for $S_{ut} > 40$ ksi (280 MPa)

Correction Factors to Endurance Limit and Fatigue Strength

 $S_e = C_{load} C_{size} C_{surf} C_{temp} C_{reliab} S_{e'}$

 $S_f = C_{load} C_{size} C_{surf} C_{temp} C_{reliab} S_{f'}$

S_e - corrected endurance limit for a part

S_f-corrected fatigue strength for a part

Correction Factors to Endurance Limit and Fatigue Strength

Cload - load factor see eq. 6.7a pg 348

C_{size} - size factor see eq's. 6.7a,b,c pg 348-349

C_{surf} - surface factor see Fig's. 6-26,
 6.27, and eq. 6.7e pg 349-352

Correction Factors to Endurance Limit and Fatigue Strength

- C_{temp} temperature factor see eq. 6.7f pg 352-353
- C_{reliab} reliability factor see Table 6.4 pg 353

Stress Concentration $K_f = 1 + q(K_t + 1)$

K_t - geometric stress concentration factor

K_f - _

q - notch sensitivity factor



Notch-Sensitivity Curves for Steels Calculated from Equation 6.13 using Data from Figure 6-35 as Originally Proposed by R. E. Peterson in "Notch Sensitivity," Chapter 13, in *Metal Fatigue* by G. Sines and J. Waisman, McGraw-Hill, New York, 1959.

Fluctuating Stresses

Modified-Goodman Diagram

Failure Lines for Fluctuating Stress



FIGURE 6-42

Various Failure Lines for Fluctuating Stresses





Stress Concentration

Apply K_f to the _____ components of stress.

For _____ materials apply K_t to the mean components of stress.

Stress Concentration - cont'd

• If
$$K_f |\sigma_{max_{nom}}| < S_y$$
 then: $K_{fm} = K_f$

If
$$K_f |\sigma_{max_{nom}}| > S_y$$
 then:
 $K_{fm} = (S_y - K_f \sigma_{anom}) / |\sigma_{mnom}|$

If
$$K_f |\sigma_{max_{nom}} - \sigma_{min_{nom}}| < 2S_y$$
 then: $K_{fm} = 0$



FOR CASE 1 failure occurs at point Q and the safety factor is the ratio of the lines YQ/YZ. To express this mathematically, we can solve equation 6.16d for the value of $\sigma'_{m@Q}$ and divide that by $\sigma'_{m@Z}$.

$$\sigma'_{m(\alpha)Q} = \left(1 - \frac{\sigma'_{a}}{S_{v}}\right) S_{v} \qquad N_{f} = \frac{\sigma'_{m(\alpha)Q}}{\sigma'_{m(\alpha)Z}} = \frac{S_{v}}{\sigma'_{m}} \left(1 - \frac{\sigma'_{a}}{S_{v}}\right) \qquad (6.18a)$$

If σ_a were so large and σ_m so small that point Q was on line CD instead of DE, then equation 6.16c should be used instead to determine the value of $\sigma_{m(aQ)}$.



FOR CASE 2 failure occurs at point P and the safety factor is the ratio of the lines XP/XZ. To express this mathematically, we can solve equation 6.16c for the value of $\sigma_{a(w)P}$ and divide that by $\sigma_{a(w)Z}$.

$$\sigma'_{a(a'P)} = \left(1 - \frac{\sigma'_{m}}{S_{ut}}\right) S_{f} \qquad N_{f} = \frac{\sigma'_{a(a'P)}}{\sigma'_{a(a'Z)}} = \frac{S_{f}}{\sigma'_{a}} \left(1 - \frac{\sigma'_{m}}{S_{ut}}\right)$$
(6.18b)

If σ'_m were so large and σ'_a so small that point *P* was on line *DE* instead of *CD*, then equation 6.16*d* should be used instead to determine the value of $\sigma'_{a(\omega P)}$.



FOR CASE 3 failure occurs at point R and the safety factor is the ratio of the lines OR/OZ or by similar triangles, either of the ratios $\sigma'_{m(\omega R)}/\sigma'_{m(\omega Z)}$ or $\sigma'_{a(\omega R)}/\sigma'_{a(\omega Z)}$. To express this mathematically, we can solve equations 6.16c and the equation of line OR simultaneously for the value of $\sigma'_{m(\omega R)}$ and divide that by $\sigma'_{m(\omega Z)}$.

from eq. 6.16c:
$$\sigma'_{a(\omega)R} = \left(1 - \frac{\sigma'_{m(\omega)R}}{S_{ut}}\right)S_f$$
(6.18c)

from line OR: $\sigma'_{a(a)R} = \left(\frac{\sigma'_{a(a)Z}}{\sigma'_{m(a)Z}}\right)\sigma'_{m(a)R} = \left(\frac{\sigma'_{a}}{\sigma'_{m}}\right)\sigma'_{m(a)R}$

The simultaneous solution of these gives

$$\sigma'_{m(a)R} = \frac{S_f}{\frac{\sigma'_a}{\sigma'_m} + \frac{S_f}{S_{ut}}}$$
(6.18d)

which after substitution and some manipulation yields

$$N_f = \frac{\sigma'_{m(a)R}}{\sigma'_{m(a)Z}} = \frac{S_f S_{ut}}{\sigma'_a S_{ut} + \sigma'_m S_f}$$
(6.18e)

There is also the possibility that point R may lie on line DE instead of CD in which case equation 6.16d should be substituted for 6.16c in the above solution.



For CASE 4 in which the future relationship between the mean and alternating stress components is either random or unknown, the point S on the failure line closest to the stress state at Z can be taken as a conservative estimate of the failure point. Line ZS is normal to CD, so its equation can be written and solved simultaneously with that of the line CD to find the coordinates of point S and the length ZS, which are

Factor of Safety for Fluctuating Stress –

$$Case \ 4$$

$$\sigma'_{m @ S} = \frac{S_{ut}(S_{f}^{2} - S_{f}\sigma'_{a} + S_{ut}\sigma'_{m})}{S_{f}^{2} + S_{ut}^{2}}$$

$$\sigma'_{a @ S} = -\frac{S_{f}}{S_{ut}}(\sigma'_{m @ S}) + S_{f}$$

$$ZS = \sqrt{(\sigma'_{m} - \sigma'_{m @ S})^{2} + (\sigma'_{a} - \sigma'_{a @ S})^{2}} \qquad (6.18f)$$

To establish a ratio for the safety factor, swing point S about point Z to be coincident with line OZR at point S'. The safety factor is the ratio OS'/OZ.

$$OZ = \sqrt{(\sigma'_{u})^{2} + (\sigma'_{m})^{2}}$$

$$N_{f} = \frac{OZ + ZS}{OZ}$$
(6.18g)

There is also the possibility that point S may lie on line DE instead of CD in which case equation 6.16d should be substituted for 6.16c in the above solution.

Fluctuating Stresses

Multiaxial Stresses in Fatigue

Fully Reversed Simple
Multiaxial Stresses

$$\sigma_{a}^{\prime} = \sqrt{\sigma_{1a}^{2} + \sigma_{2a}^{2} + \sigma_{3a}^{2} - \sigma_{1a}\sigma_{2a} - \sigma_{2a}\sigma_{3a} - \sigma_{1a}\sigma_{3a}} \qquad (6.19a)$$

and for the two-dimensional case:

$$\sigma'_{a} = \sqrt{\sigma_{1_{a}}^{2} - \sigma_{1_{a}}\sigma_{2_{a}} + \sigma_{2_{a}}^{2}}$$
(6.19b)

$$N_f = \frac{S_n}{\sigma_a}$$

Fluctuating Simple Multiaxial Stresses

Von Mises Method

$$\sigma'_{a} = \sqrt{\frac{\left(\sigma_{x_{a}} - \sigma_{y_{a}}\right)^{2} + \left(\sigma_{y_{a}} - \sigma_{z_{a}}\right)^{2} + \left(\sigma_{z_{a}} - \sigma_{x_{a}}\right)^{2} + 6\left(\tau_{xy_{a}}^{2} + \tau_{yz_{a}}^{2} + \tau_{zx_{a}}^{2}\right)}{2}}{(6.22a)}$$
$$\sigma'_{m} = \sqrt{\frac{\left(\sigma_{x_{m}} - \sigma_{y_{m}}\right)^{2} + \left(\sigma_{y_{m}} - \sigma_{z_{m}}\right)^{2} + \left(\sigma_{z_{m}} - \sigma_{x_{m}}\right)^{2} + 6\left(\tau_{xy_{m}}^{2} + \tau_{yz_{m}}^{2} + \tau_{zx_{m}}^{2}\right)}{2}}{2}$$

or for a biaxial stress state using:

$$\sigma'_{a} = \sqrt{\sigma_{x_{a}}^{2} + \sigma_{y_{a}}^{2} - \sigma_{x_{a}}\sigma_{y_{a}} + 3\tau_{xy_{a}}^{2}}$$

$$\sigma'_{m} = \sqrt{\sigma_{x_{m}}^{2} + \sigma_{y_{m}}^{2} - \sigma_{x_{m}}\sigma_{y_{m}} + 3\tau_{xy_{m}}^{2}}$$
(6)

.22b)

General Approach to High Cycle Fatigue Design

- Generate Modified Goodman Diagram
- Calculate alternating and mean components of stress at areas of concern on the part. Include appropriate stress concentration factors.
- Convert alternating and mean applied stresses to alternating and mean Von Mises Stresses.
- Plot these stresses on the Modified Goodman Diagram and find the factor of safety.

Example

- The figure pertains to the shaft of a disk sander that is made from steel having Su=900 MPa, and Sy=750 MPa. The most severe loading occurs when an object is held near the periphery of the disk with sufficient force to develop 12 N-m (which approaches the stall torque of the motor).
- Assume a coefficient of friction between the object and the disk is 0.6.
- What is the factor of safety with respect to eventual fatigue failure of the shaft?

Example

Schematic and Given Data:

