

One-Dimensional, Steady-State
Conduction without
Thermal Energy Generation

Methodology of a Conduction Analysis

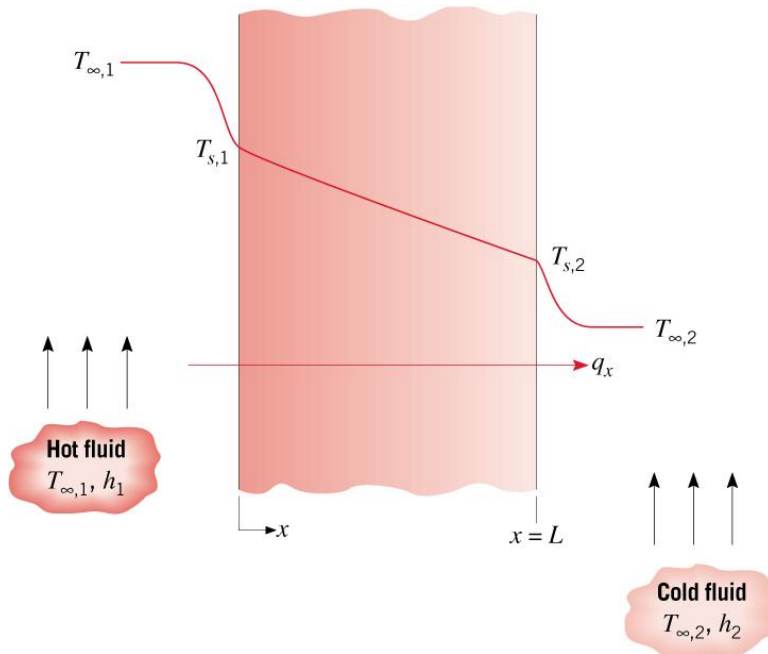
- Specify appropriate form of the **heat equation**.
- Solve for the **temperature distribution**.
- Apply **Fourier's Law** to determine the **heat flux**.

Simplest Case: **One-Dimensional, Steady-State** Conduction with **No Thermal Energy Generation**.

- Common Geometries:
 - The **Plane Wall**: Described in rectangular (x) coordinate. Area perpendicular to direction of heat transfer is constant (independent of x).
 - The **Tube Wall**: Radial conduction through tube wall.
 - The **Spherical Shell**: Radial conduction through shell wall.

The Plane Wall

- Consider a plane wall between two fluids of different temperature:



- Heat Equation:**

$$\frac{d}{dx} \left(k \frac{dT}{dx} \right) = 0 \quad (3.1)$$

- Implications:**

Heat flux (q_x'') is independent of x .

Heat rate (q_x) is independent of x .

- Boundary Conditions:** $T(0) = T_{s,1}$, $T(L) = T_{s,2}$

- Temperature Distribution** for Constant k :

$$T(x) = T_{s,1} + (T_{s,2} - T_{s,1}) \frac{x}{L} \quad (3.3)$$

- **Heat Flux and Heat Rate:**

$$q_x'' = -k \frac{dT}{dx} = \frac{k}{L} (T_{s,1} - T_{s,2}) \quad (3.5)$$

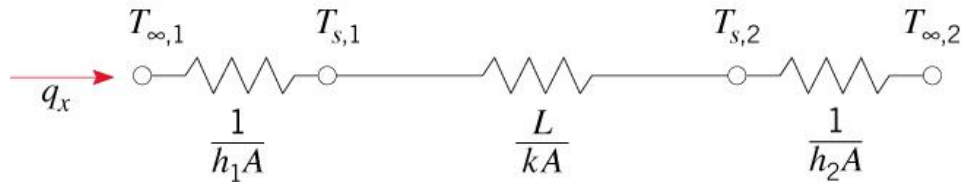
$$q_x = -kA \frac{dT}{dx} = \frac{kA}{L} (T_{s,1} - T_{s,2}) \quad (3.4)$$

- **Thermal Resistances** $\left(R_t = \frac{\Delta T}{q} \right)$ **and Thermal Circuits:**

Conduction in a plane wall: $R_{t,cond} = \frac{L}{kA}$ (3.6)

Convection: $R_{t,conv} = \frac{1}{hA}$ (3.9)

Thermal circuit for plane wall with adjoining fluids:



$$R_{tot} = \frac{1}{h_1 A} + \frac{L}{kA} + \frac{1}{h_2 A} \quad (3.12)$$

$$q_x = \frac{T_{\infty,1} - T_{\infty,2}}{R_{tot}} \quad (3.11)$$

- Thermal Resistance for **Unit Surface Area**:

$$R''_{t,cond} = \frac{L}{k} \quad R''_{t,conv} = \frac{1}{h}$$

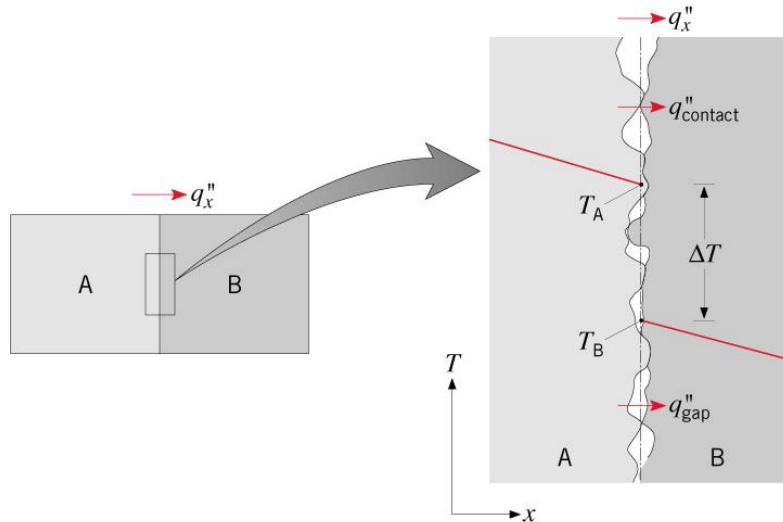
Units: $R_t \leftrightarrow \text{W/K}$ $R''_t \leftrightarrow \text{m}^2 \cdot \text{K/W}$

- **Radiation Resistance:**

$$R_{t,rad} = \frac{1}{h_r A} \quad R''_{t,rad} = \frac{1}{h_r}$$

$$h_r = \varepsilon \sigma (T_s + T_{sur}) (T_s^2 + T_{sur}^2) \quad (1.9)$$

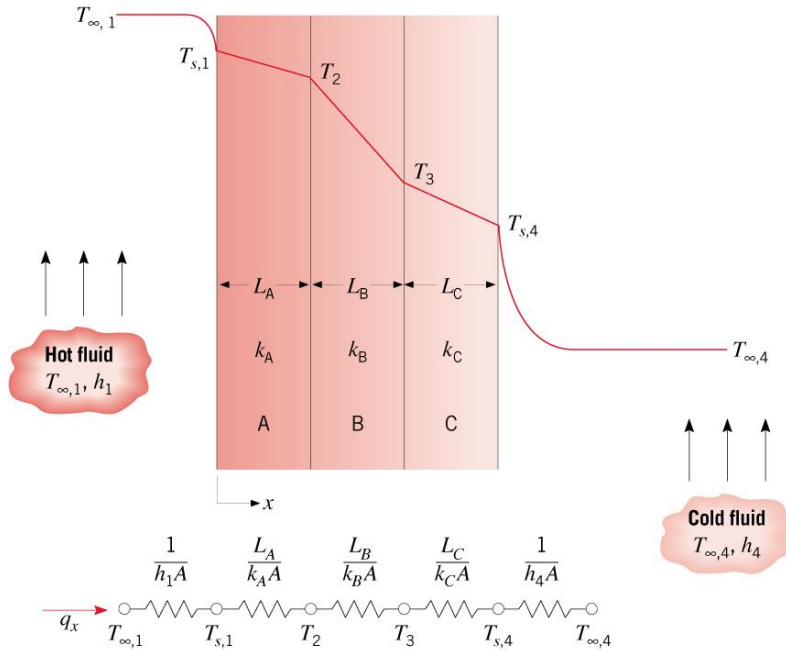
- **Contact Resistance:**



$$R''_{tc} = \frac{T_A - T_B}{q''_x} \quad R_{t,c} = \frac{R''_{t,c}}{A_c}$$

Values depend on: Materials A and B, surface finishes, interstitial conditions, and contact pressure (Tables 3.1 and 3.2)

Plane Wall (cont.)



- Composite Wall with Negligible Contact Resistance:

$$q_x = \frac{T_{\infty,1} - T_{\infty,4}}{R_{tot}} \quad (3.14)$$

$$R_{tot} = \frac{1}{A} \left[\frac{1}{h_1} + \frac{L_A}{k_A} + \frac{L_B}{k_B} + \frac{L_C}{k_C} + \frac{1}{h_4} \right] = \frac{R''_{tot}}{A}$$

- Overall Heat Transfer Coefficient (U):

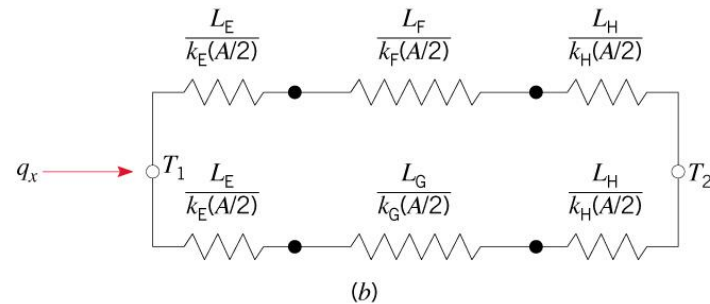
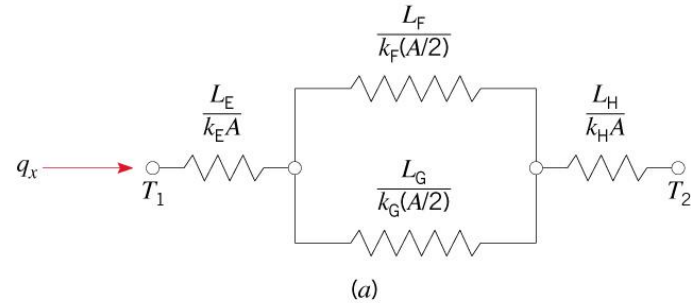
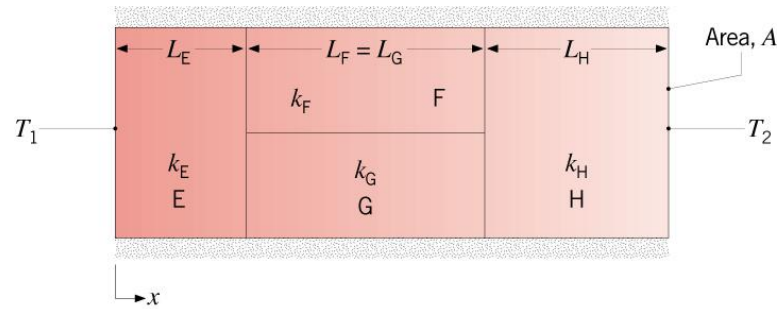
A modified form of Newton's Law of Cooling to encompass multiple resistances to heat transfer.

$$q_x = UA\Delta T_{overall} \quad (3.17)$$

$$R_{tot} = \frac{1}{UA} \quad (3.19)$$

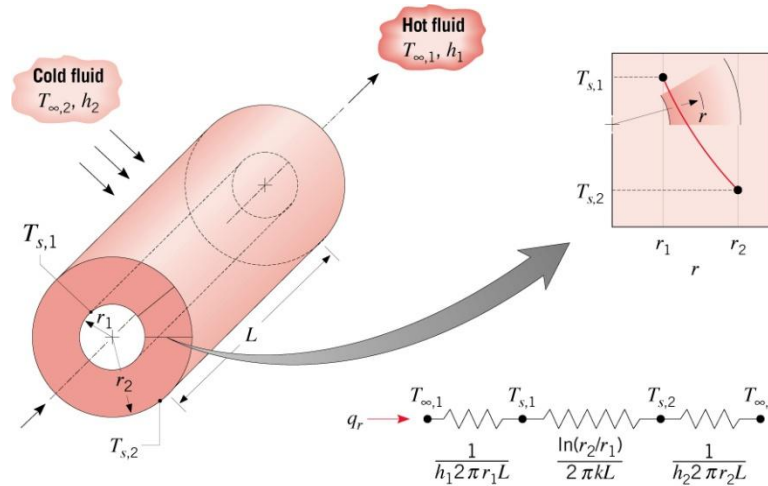
Plane Wall (cont.)

- Series – Parallel Composite Wall:



- Note departure from one-dimensional conditions for $k_F \neq k_G$.
- Circuits based on assumption of isothermal surfaces normal to x direction or adiabatic surfaces parallel to x direction provide approximations for q_x .

The Tube Wall



- Heat Equation:

$$\frac{1}{r} \frac{d}{dr} \left(kr \frac{dT}{dr} \right) = 0 \tag{3.23}$$

What does the form of the heat equation tell us about the variation of q_r with r in the wall?

Is the foregoing conclusion consistent with the energy conservation requirement?

How does q_r'' vary with r ?

- Temperature Distribution for Constant k :

$$T(r) = \frac{T_{s,1} - T_{s,2}}{\ln(r_1/r_2)} \ln\left(\frac{r}{r_2}\right) + T_{s,2} \tag{3.26}$$

- **Heat Flux** and **Heat Rate**:

$$q_r'' = -k \frac{dT}{dr} = \frac{k}{r \ln(r_2/r_1)} (T_{s,1} - T_{s,2})$$

$$q_r' = 2\pi r q_r'' = \frac{2\pi k}{\ln(r_2/r_1)} (T_{s,1} - T_{s,2})$$

$$q_r = 2\pi r L q_r'' = \frac{2\pi L k}{\ln(r_2/r_1)} (T_{s,1} - T_{s,2}) \quad (3.27)$$

- **Conduction Resistance**:

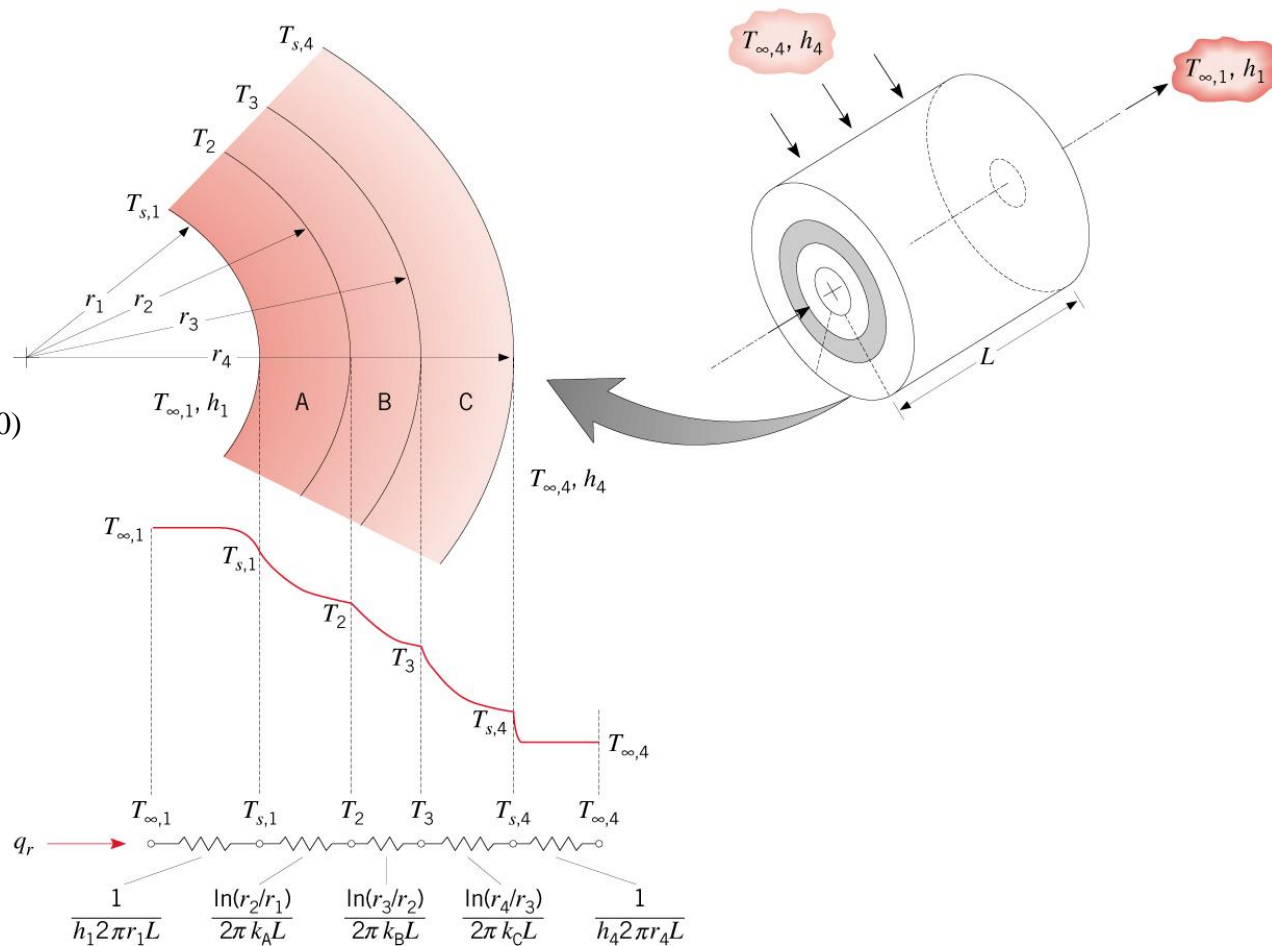
$$R_{t,cond} = \frac{\ln(r_2/r_1)}{2\pi L k} \quad \text{Units} \leftrightarrow \text{K/W} \quad (3.28)$$

$$R'_{t,cond} = \frac{\ln(r_2/r_1)}{2\pi k} \quad \text{Units} \leftrightarrow \text{m} \cdot \text{K/W}$$

Why is it inappropriate to base the thermal resistance on a unit surface area?

- **Composite Wall with Negligible Contact Resistance**

$$q_r = \frac{T_{\infty,1} - T_{\infty,4}}{R_{tot}} = UA(T_{\infty,1} - T_{\infty,4}) \quad (3.30)$$



Note that

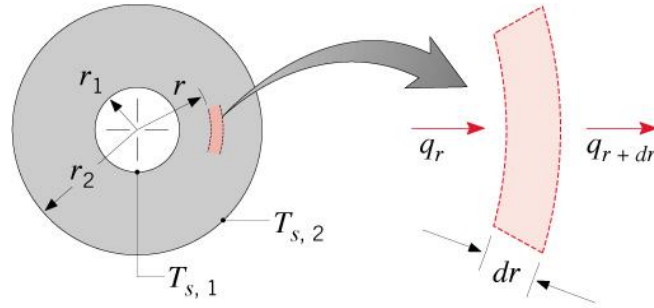
$$UA = R_{tot}^{-1}$$

is a constant independent of radius.

But, U itself is tied to specification of an interface.

$$U_i = (A_i R_{tot})^{-1} \quad (3.32)$$

Spherical Shell



- Heat Equation**

$$\frac{1}{r^2} \frac{d}{dr} \left(r^2 \frac{dT}{dr} \right) = 0$$

What does the form of the heat equation tell us about the variation of q_r with r ? Is this result consistent with conservation of energy?

How does q_r'' vary with r ?

- Temperature Distribution** for Constant k :

$$T(r) = T_{s,1} - (T_{s,1} - T_{s,2}) \frac{1 - (r_1/r)}{1 - (r_1/r_2)}$$

- **Heat flux, Heat Rate and Thermal Resistance:**

$$q_r'' = -k \frac{dT}{dr} = \frac{k}{r^2 \left[(1/r_1) - (1/r_2) \right]} (T_{s,1} - T_{s,2})$$

$$q_r = 4\pi r^2 q_r'' = \frac{4\pi k}{(1/r_1) - (1/r_2)} (T_{s,1} - T_{s,2}) \quad (3.35)$$

$$R_{t,cond} = \frac{(1/r_1) - (1/r_2)}{4\pi k} \quad (3.36)$$

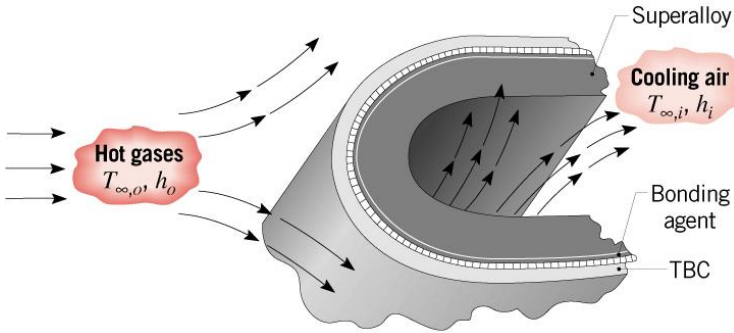
- **Composite Shell:**

$$q_r = \frac{\Delta T_{overall}}{R_{tot}} = UA \Delta T_{overall}$$

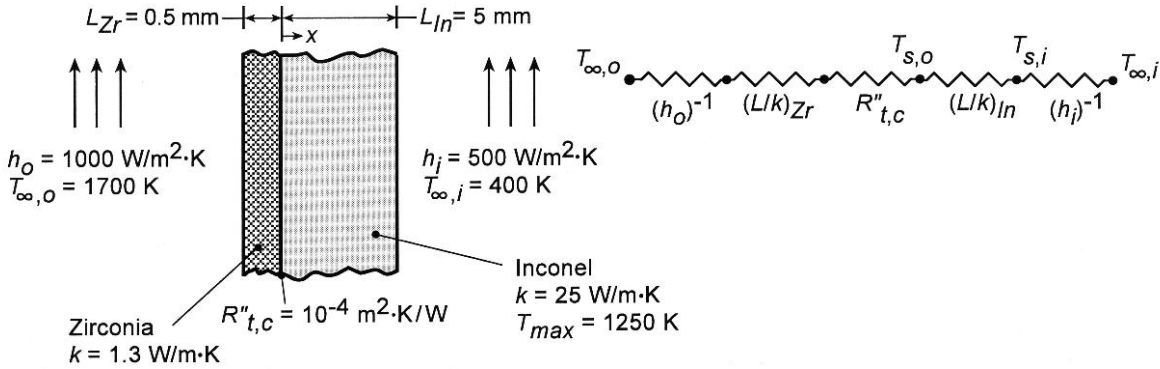
$$UA = R_{tot}^{-1} \leftrightarrow \text{Constant}$$

$$U_i = (A_i R_{tot})^{-1} \leftrightarrow \text{Depends on } A_i$$

Problem 3.23: Assessment of thermal barrier coating (TBC) for protection of turbine blades. Determine maximum blade temperature with and without TBC.



Schematic:



ASSUMPTIONS: (1) One-dimensional, steady-state conduction in a composite plane wall, (2) Constant properties, (3) Negligible radiation.

ANALYSIS: For a unit area, the total thermal resistance with the TBC is

$$R''_{tot,w} = h_o^{-1} + (L/k)_{Zr} + R''_{t,c} + (L/k)_{In} + h_i^{-1}$$

$$R''_{tot,w} = \left(10^{-3} + 3.85 \times 10^{-4} + 10^{-4} + 2 \times 10^{-4} + 2 \times 10^{-3}\right) m^2 \cdot K/W = 3.69 \times 10^{-3} m^2 \times K/W$$

With a heat flux of

$$q_w'' = \frac{T_{\infty,o} - T_{\infty,i}}{R''_{tot,w}} = \frac{1300 \text{ K}}{3.69 \times 10^{-3} m^2 \times K/W} = 3.52 \times 10^5 \text{ W/m}^2$$

the inner and outer surface temperatures of the Inconel are

$$T_{s,i(w)} = T_{\infty,i} + (q_w''/h_i) = 400 \text{ K} + \left(3.52 \times 10^5 \text{ W/m}^2 / 500 \text{ W/m}^2 \times \text{K}\right) = 1104 \text{ K}$$

$$T_{s,o(w)} = T_{\infty,i} + \left[(1/h_i) + (L/k)_{In}\right] q_w'' = 400 \text{ K} + \left(2 \times 10^{-3} + 2 \times 10^{-4}\right) m^2 \times K/W \left(3.52 \times 10^5 \text{ W/m}^2\right) = 1174 \text{ K}$$

Without the TBC,

$$R''_{tot,wo} = h_o^{-1} + (L/k)_{In} + h_i^{-1} = 3.20 \times 10^{-3} \text{ m}^2 \times \text{K/W}$$

$$q''_{wo} = (T_{\infty,o} - T_{\infty,i}) / R''_{tot,wo} = 4.06 \times 10^5 \text{ W/m}^2.$$

the inner and outer surface temperatures of the Inconel are

$$T_{s,i(wo)} = T_{\infty,i} + (q''_{wo} / h_i) = 1212 \text{ K}$$

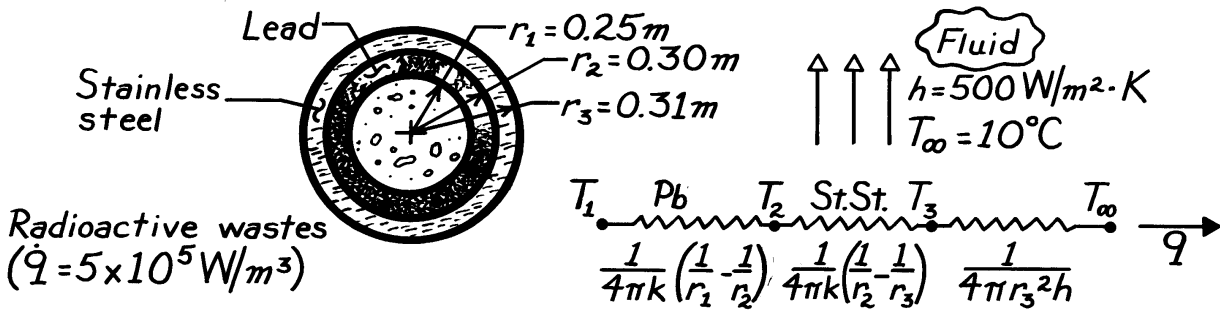
$$T_{s,o(wo)} = T_{\infty,i} + \left[(1/h_i) + (L/k)_{In} \right] q''_{wo} = 1293 \text{ K}$$

Use of the TBC facilitates operation of the Inconel below $T_{\max} = 1250 \text{ K}$.

COMMENTS: Since the durability of the TBC decreases with increasing temperature, which increases with increasing thickness, limits to its thickness are associated with reliability considerations.

Problem 3.62: Suitability of a composite spherical shell for storing radioactive wastes in oceanic waters.

SCHEMATIC:



ASSUMPTIONS: (1) One-dimensional conduction, (2) Steady-state conditions, (3) Constant properties at 300K, (4) Negligible contact resistance.

PROPERTIES: Table A-1, Lead: $k = 35.3 \text{ W/m}\cdot\text{K}$, $MP = 601\text{K}$; St.St.: $15.1 \text{ W/m}\cdot\text{K}$.

ANALYSIS: From the thermal circuit, it follows that

$$q = \frac{T_1 - T_\infty}{R_{\text{tot}}} = \dot{q} \left[\frac{4}{3} \pi r_1^3 \right]$$

The thermal resistances are:

$$R_{Pb} = \left[1 / (4\pi \times 35.3 \text{ W/m} \cdot \text{K}) \right] \left[\frac{1}{0.25\text{m}} - \frac{1}{0.30\text{m}} \right] = 0.00150 \text{ K/W}$$

$$R_{St.St.} = \left[1 / (4\pi \times 15.1 \text{ W/m} \cdot \text{K}) \right] \left[\frac{1}{0.30\text{m}} - \frac{1}{0.31\text{m}} \right] = 0.000567 \text{ K/W}$$

$$R_{conv} = \left[1 / \left(4\pi \times 0.31^2 \text{ m}^2 \times 500 \text{ W/m}^2 \cdot \text{K} \right) \right] = 0.00166 \text{ K/W}$$

$$R_{tot} = 0.00372 \text{ K/W.}$$

The heat rate is then

$$q = 5 \times 10^5 \text{ W/m}^3 (4\pi/3)(0.25\text{m})^3 = 32,725 \text{ W}$$

and the inner surface temperature is

$$T_1 = T_\infty + R_{tot} q = 283\text{K} + 0.00372\text{K/W} (32,725 \text{ W}) = 405 \text{ K} < MP = 601\text{K.}$$

Hence, from the thermal standpoint, the proposal is adequate.

COMMENTS: In fabrication, attention should be given to maintaining a good thermal contact. A protective outer coating should be applied to prevent long term corrosion of the stainless steel.